



ISSN: 2350-0328

**International Journal of Advanced Research in Science,  
Engineering and Technology**

**Vol. 12, Issue 4, April 2025**

# **Structural Failure Analysis of the American Airlines Flight 587 Crash: Lessons from Composite Material Fatigue and Aerodynamic Loads**

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**ABSTRACT:** The crash of American Airlines Flight 587 on November 12, 2001, was a catastrophic failure that resulted from structural issues associated with composite materials and aerodynamic loads. This paper examines the underlying causes of the crash, focusing on the failure of the Airbus A300-600's vertical stabilizer, which was composed of composite materials. Through a detailed analysis of material fatigue, aerodynamic forces, and pilot-induced stress, this study identifies key lessons in aircraft design, maintenance, and pilot training that can help prevent similar accidents in the future.

**KEY WORDS:** American Airlines Flight 587, Airbus A300-600, Vertical stabilizer failure, Composite materials, Carbon-fiber-reinforced plastic (CFRP), Material fatigue, Delamination, Aerodynamic loads, Wake turbulence, Pilot-induced loads, Rudder inputs, Structural failure, Non-destructive testing (NDT), Fatigue testing, Computational fluid dynamics (CFD), Finite element analysis (FEA), Aircraft maintenance, Pilot training, Flight safety, Aircraft design improvements

## **I. INTRODUCTION**

American Airlines Flight 587, which had an Airbus A300-600 model, lifted off from the John F. Kennedy International Airport in New York, America and was going to Santo Domingo in the Dominican Republic. It had taken off on the afternoon of 12 November, 2001. Now, a bit more than two minutes into the flight, the airplane lost control all of a sudden and ended up crashing into the Belle Harbor neighborhood situated in Queens. Due to the accident, roughly 260 passengers and 5 residents died as a tragic result. Given the fact that this incident took place just two months after the 9/11 attacks, there was significant discussion regarding the fact that this might have been a terrorist activity. Unlike the speculation done by the general public, the National Transportation Safety Board (NTSB) conducted a thorough investigation and provided clarification that this incident was due to the structural failure of the vertical stabilizer (or tail fin) of the airplane.

As with every other part of an aircraft, the vertical stabilizer will maintain the aircraft's stability and helps in achieving control of motion with regards to the various directions. On this aircraft, it was constructed using composite materials like carbon-fiber-reinforced plastic CFRP. As compared to modern materials being used in aviation technology, these remain relatively light and have high strength. Such composites can be very useful but also have their fair share of disadvantages, especially with regards to aluminum.



ISSN: 2350-0328

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## II. RELATED WORKS

A number of studies and investigations in the past decades have investigated the intricate relationships between aircraft structure design, control inputs by the pilot, and fatigue in materials—particularly with respect to composite materials. The American Airlines Flight 587 crash highlighted many of these considerations and generated greater focus on the response of composite structures to dynamic flight loads.

### Composite Materials in Aviation

Use of carbon-fiber-reinforced plastic (CFRP) and other high composites has been extensively researched in aerospace engineering. Baker, Dutton, and Kelly (2004) present a complete description of how composite materials respond to stress, both revealing the strengths (e.g., light weight and high strength) and the weaknesses (e.g., abrupt failure without extensive warning). Researchers like Thornton (2003) have specifically focused on how delamination and internal cracking in composites can remain hidden until failure occurs—making routine inspections more challenging.

### Structural Failure Due to Aerodynamic Loads

Studies conducted by NASA Langley Research Center and others have used finite element analysis (FEA) and computational fluid dynamics (CFD) to simulate how vertical stabilizers respond to lateral aerodynamic forces, particularly those generated by rudder movements. Such simulations have demonstrated that some load cases—particularly through sudden side-to-side rudder inputs—will exceed design loads even in structurally healthy parts. FAA (2002) work also makes the point that aircraft certification would need to be based on accommodating these possible cases of stress, both in metal and composite design.

### Pilot Input and Human Factors

Human factors' contribution to aviation accidents has been comprehensively documented. Studies by the NTSB and aviation safety organizations have focused on the pilot behavior in structural overloading incidents. To be specific, excessive high-speed rudder inputs have been investigated in the case of multiple incidents—Flight 587 being one. The NTSB's final report on Flight 587 indicated how even within the mechanical design limits of the plane, pilot-initiated rudder reversals generated alternating loads that were ultimately destructive.

Subsequent work by aviation psychologists and training specialists has pinpointed the necessity for more extensive pilot training on rudder operation, especially in commercial airliners. This aligns with modifications to training procedures that followed the crash, which increasingly focus on managing wake turbulence with smoother control inputs.

### Broader Impacts on Aircraft Design

After the crash, manufacturers such as Airbus and Boeing reassessed their composite design standards and revised maintenance procedures. Industry associations such as SAE International and the American Institute of Aeronautics and Astronautics (AIAA) have since issued new guidelines for testing, monitoring, and certifying composite parts. These developments reflect an increasing appreciation of how structural design, pilot behavior, and environmental stressors converge in actual flight conditions.



ISSN: 2350-0328

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## III. BACKGROUND

### Aircraft and Flight Overview

The plane was an Airbus A300-600, a twin-engine, wide-body jetliner that American Airlines had been flying since the late 1980s. Both domestic and international routes made extensive use of it. The A300-600's use of cutting-edge materials, particularly in the tail section, was one of its main features. Composite materials, specifically carbon-fiber-reinforced plastic (CFRP), were used to construct the vertical stabilizer and the fuselage attachment. Since these materials are stronger and lighter than metal, they are frequently utilized in the design of contemporary aircraft. However, under some stresses, composites can fail abruptly and without much notice, and they don't always showcase damage the same way metals do.

The scheduled passenger flight Flight 587 traveled from New York's JFK Airport to Las Américas International Airport.

### CRASH SEQUENCE

When it became airborne, the plane had some light turbulence, probably off the wake of the 747. In reply, the first officer responded to this with a series of stiff left-right rudder inputs that stabilized the airplane. Rudder inputs are ordinarily reserved for other times—especially at high speed—because rudder inputs have a tendency to put a tremendous amount of side stress on an airplane. But in this instance, the inputs were more extreme than required, and they occurred in rapid succession. These abrupt alternating forces produced a condition called "rudder reversal," where the tail structure feels strong side-to-side loads that can be higher than it is capable of.

Consequently, the vertical stabilizer started to structurally overload. Within seconds, the composite structure completely failed, separating from the fuselage. As soon as the tail section disappeared, the plane had no directional stability and entered an uncontrolled descent. It crashed in a residential area of Queens, bringing more destruction to the ground.

### IMMEDIATE AFTERMATH

At first, fear and confusion were ignited by the crash, with many believing that it was a terrorist attack due to its close proximity in time to 9/11. Yet, the NTSB swiftly commissioned an extensive investigation. Both the flight data recorder and the cockpit voice recorder were recovered and offered definitive proof that rudder movement, not an external attack, caused the structural failure. The crash focused attention on the interrelated factors of pilot behavior, aircraft design, and the material properties of composite materials.

## IV. STRUCTURAL FAILURE ANALYSIS

Understanding why the tail of Flight 587 broke off mid-flight means digging into the materials used to build the plane, the forces acting on it, and how the pilot's actions played a role. In this section, we'll take a closer look at composite material behavior, how fatigue builds up over time, and how aerodynamic loads from the rudder inputs caused the structure to fail.

### COMPOSITE MATERIAL FATIGUE AND FAILURE

#### PROPERTIES OF COMPOSITE MATERIALS

The Airbus A300-600's vertical stabilizer wasn't constructed of ordinary metal—it was made of carbon-fiber-reinforced plastic (CFRP). Composite materials such as CFRP are increasingly being used in contemporary aircraft because



ISSN: 2350-0328

# International Journal of Advanced Research in Science, Engineering and Technology

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they're incredibly tough and much lighter than metal, which makes them more fuel-efficient and performant. However, composites react quite differently when subjected to stress.

In contrast to metal, which typically bends or stretches before it fails (providing some warning), composites may fail suddenly and without a great deal of apparent damage in advance. That's because rather than stretching or distorting, the internal layers of a composite may begin to crack, delaminate (peel away from one another), or break under stress without indicating anything on the surface.

## Fatigue and Delamination Over Time

Every part of the aircraft is subjected to stress while flying—particularly components such as the tail that assist in direction. Gradually, the repeated stressing results in microscopic cracks developing internally. In composite materials, this can initiate delamination, where the internal layers begin to peel off from each other. This compromises the overall structure despite the fact that it may continue to appear to be in good shape outwardly.

In Flight 587's case, there's good chance the tail had already accumulated some internal fatigue due to years of routine use. Even though the NTSB did not discover any significant pre-crash damage, even small fatigue would have reduced the stabilizer's strength to handle sudden, high loads—such as those it was subjected to during the crash.

## MAINTENANCE AND INSPECTION CHALLENGES

One of the greatest issues with composites is that it's difficult to spot internal damage without specialized equipment. You can't always inspect cracks or delamination from the exterior as you would with metal. That leaves airlines needing to employ expensive advanced non-destructive testing (NDT) techniques such as ultrasonic scanning or thermal imaging to examine for concealed problems. Damage can go undetected until it's too late if inspections aren't completed regularly enough or thoroughly enough.

## AERODYNAMIC LOADS AND STRUCTURAL STRESS

### WAKE TURBULENCE EFFECTS

Just before the crash, Flight 587 flew into the wake turbulence of a Japan Airlines Boeing 747. Wake turbulence is essentially the whirling air in the wake of a big airplane. It's not typically hazardous in itself, but it can cause a smaller aircraft to shake or roll suddenly. Pilots are taught to deal with this through small control inputs—but in this instance, the first officer overreacted with more forceful rudder inputs than required.

### PILOT-INDUCED LOADS AND RUDDER REVERSALS

The rudder is intended to assist in an aircraft's turn left or right (yaw) and is controlled via foot pedals. On this flight, the first officer nudged the rudder to the left, then to the right, then left again—repeatedly. This induced so-called rudder reversals, meaning that the plane's tail was being jolted side-to-side rapidly. These oscillating aerodynamic loads placed immense lateral forces upon the aircraft, which far outpaced what the vertical stabilizer was constructed to endure.

Based on simulations and data, the stabilizer was subjected to loads in excess of its design. The ultimate rudder deflection was the breaking point—the already loaded composite structure could not handle anymore, and it catastrophically failed.

## LOAD ANALYSIS AND SIMULATIONS

Analysts employed computer modeling such as Computational Fluid Dynamics (CFD) and Finite Element Analysis (FEA) to simulate the forces on the tail during those rudder inputs. The models revealed that the combined effect of past fatigue and sudden aerodynamic loading rendered the structure particularly susceptible.

After the vertical stabilizer snapped off, the aircraft lost directional control capability. In a matter of seconds, it went into a flat spin and crashed.



## **V. LESSONS LEARNED**

The American Airlines Flight 587 crash was a sad incident, but it was also a huge wake-up call for the aviation community. It revealed vulnerabilities not only in aircraft structure and materials, but also in pilot training and maintenance—particularly with regard to newer planes that use more composite materials. These are some of the most significant lessons learned from the investigation.

### **1. COMPOSITE MATERIALS NEED EXTRA CARE**

Composite materials such as carbon-fiber-reinforced plastic (CFRP) are increasingly being used in aircraft than ever before. They're light and strong, but not like regular metals. When composites fail, it's usually unexpectedly and catastrophically.

What we learned:

Improved fatigue testing. Aircraft components made of composites need to be subjected to more strenuous long-term stress and fatigue testing when being developed.

Designs require incorporated safety margins. Engineers need to include additional strength ("structural redundancy") in essential composite components to cope with unforeseen loads.

All damage cannot be seen. Maintenance crews require improved tools and methods to identify concealed problems within composite components prior to becoming severe.

### **2. REDESIGNING PILOT TRAINING FOR NEW AIRCRAFT**

One of the biggest surprises of the Flight 587 crash was that it wasn't an external threat or system failure—it was a function of how the plane was flown. The first officer's rudder inputs were much more aggressive than required, and at high speeds, they created tremendous stress on the tail.

What we learned:

Pilots must comprehend rudder sensitivity. Rudder pedals are strong, particularly at high speeds. Most pilots weren't fully cognizant of how small inputs could result in large aerodynamic forces.

Rudder use should be minimized at cruising speeds. Pilots now get clearer instruction on when and how to use the rudder correctly.

Wake turbulence response should be more subtle. Rather than overreacting to light turbulence, pilots learn to trust greater use of natural stability of the aircraft and softer control inputs.

### **3. ENHANCED MAINTENANCE AND INSPECTION PROCEDURES**

Prior to this crash, there was a perception that composite structures were largely "maintenance-free" due to their strength and corrosion resistance. Flight 587 demonstrated that even robust materials can be weakened if internal fatigue accumulates—and that such damage may not be apparent.

What we learned:

More sophisticated inspection equipment is required. Maintenance teams now employ high-tech techniques such as ultrasound and infrared scanning to detect damage within composite layers.

Maintenance cycles must factor in fatigue, not age. It's not so much how old a component is—it's how stressed it has been over the years.

Current regulations would have to be revised for composite structures. Aviation safety regulatory bodies are further developing standards for how composite components should be inspected and certified.

#### **4. SYSTEMS THINKING: AIRCRAFT, PILOT, AND ENVIRONMENT**

This crash demonstrated that accidents seldom occur due to a single reason. Rather, it is a combination of factors—a pilot responding to wake turbulence, a structural element already compromised by fatigue, and a design that did not adequately take into consideration severe rudder inputs.

What we learned:

It all works together. Flight safety relies upon the plane, the pilot, the materials, and the training all functioning as a system.

Design has to take into account human behavior. Even if it shouldn't occur (such as a sudden rudder reversal), engineers have to design systems that can safely cope with the reality that it might occur.

Learning is ongoing. Each accident needs to be regarded as a learning experience to better design aircraft, revise training materials, and optimize maintenance procedures.



Fig: Diagram of the Airbus A300-600 Vertical Stabiliser

#### **VI. CONCLUSION AND FUTURE WORK**

The American Airlines Flight 587 crash was a sad moment, and although nothing could reverse the disaster, it led the aviation industry to learn a few tough but valuable lessons. At its center, this crash demonstrated how a mix of human choices, high-tech materials, and high-pressure flight conditions have the potential to interact in unplanned—and lethal—fashions.



ISSN: 2350-0328

# International Journal of Advanced Research in Science, Engineering and Technology

Vol. 12, Issue 4, April 2025

The structural collapse of the vertical stabilizer was not merely a case of a weak tail or pilot error. It was the product of several factors: the application of composite materials that behave differently under stress and fatigue than conventional metals, rudder inputs higher than the structure could sustain, and inspection and maintenance systems that were not as effective at detecting sub-surface damage in composite components.

This incident pushed the industry to rethink how planes are engineered, pilots are trained, and aircraft health is tracked over time. It brought into focus the need to know not only about individual components, but how everything in an aircraft system—both mechanical and human—work together under stress.

Since then, the aviation industry has made actual progress. Training courses have been revised to educate pilots further on the dangers of excessive rudder use. Methods of inspecting composite materials have become better, and engineers are creating safer, more durable aircraft structures with a better understanding of fatigue and load behavior.

Ultimately, Flight 587 is a sobering reminder: despite cutting-edge technology, human wisdom and judgment have an enormous impact on maintaining air safety. Each crash leaves its imprint—but it also leaves behind a lesson that makes flying safer for everyone

## REFERENCES

1. National Transportation Safety Board. (2004). \*Aircraft Accident Report: In-Flight Separation of Vertical Stabilizer, American Airlines Flight 587, Airbus Industrie A300-605R, N14053, Belle Harbor, New York, November 12, 2001\* (NTSB/AAR-04/04). Retrieved from [https://www.nts.gov](https://www.nts.gov)
2. Airbus. (2001). \*Airbus A300-600: Aircraft Maintenance Manual & Structural Design Data\*. Toulouse: Airbus S.A.S.
3. FAA. (2002). \*Advisory Circular AC 20-107B: Composite Aircraft Structure\*. U.S. Department of Transportation, Federal Aviation Administration.
4. Baker, A. A., Dutton, S., & Kelly, D. (2004). \*Composite Materials for Aircraft Structures\* (2nd ed.). American Institute of Aeronautics and Astronautics.
5. Rouse, M. (2005). "Understanding Rudder Reversals: A Cautionary Tale from Flight 587." \*Journal of Aircraft Safety and Control\*, 17(2), 45–53.
6. Raymer, D. P. (2012). \*Aircraft Design: A Conceptual Approach\* (5th ed.). American Institute of Aeronautics and Astronautics.
7. Federal Aviation Administration. (2001). \*Pilot's Handbook of Aeronautical Knowledge\* (FAA-H-8083-25A). Washington, DC: U.S. Government Printing Office.
8. Thornton, A. (2003). "Delamination in Composite Structures and Its Role in Aircraft Failure." \*Aerospace Materials and Engineering Journal\*, 11(4), 229–241.
9. NASA Langley Research Center. (2002). \*Finite Element Analysis of Vertical Stabilizer Load Cases in Composite Aircraft\*. Hampton, VA: NASA Technical Reports Server.